

Final Exam solutions

$$1. a) -bv^{3/2} = m \frac{dv}{dt} \quad \int -\frac{dv}{v^{3/2}} = \int \frac{b}{m} dt$$

$$2v^{-1/2} \Big|_{v_i}^{v_f} = \alpha t \quad \alpha = b/m.$$

$$\frac{2}{\sqrt{v_f}} - \frac{2}{\sqrt{v_i}} = \alpha t \quad \text{if } v_f = 0 \quad t \rightarrow \infty$$

$$b) v_f = v(t)$$

$$\frac{2}{\sqrt{v_f}} = \alpha t + \frac{2}{\sqrt{v_i}}$$

$$\frac{1}{\sqrt{v_f}} = \frac{\alpha}{2} t + \frac{1}{\sqrt{v_i}} = \frac{\alpha}{2} t + \beta$$

$$\frac{1}{v_f} = \frac{dx}{dt} = \left(\frac{\alpha}{2} t + \beta\right)^2$$

$$\text{so } \int dx = \int \frac{dt}{\left(\frac{\alpha}{2} t + \beta\right)^2}$$

$$\Delta x = -\frac{1}{\frac{\alpha}{2} \left(\frac{\alpha}{2} t + \beta\right)} \Big|_{t=0}^{t=\infty}$$

from Eqn sheet

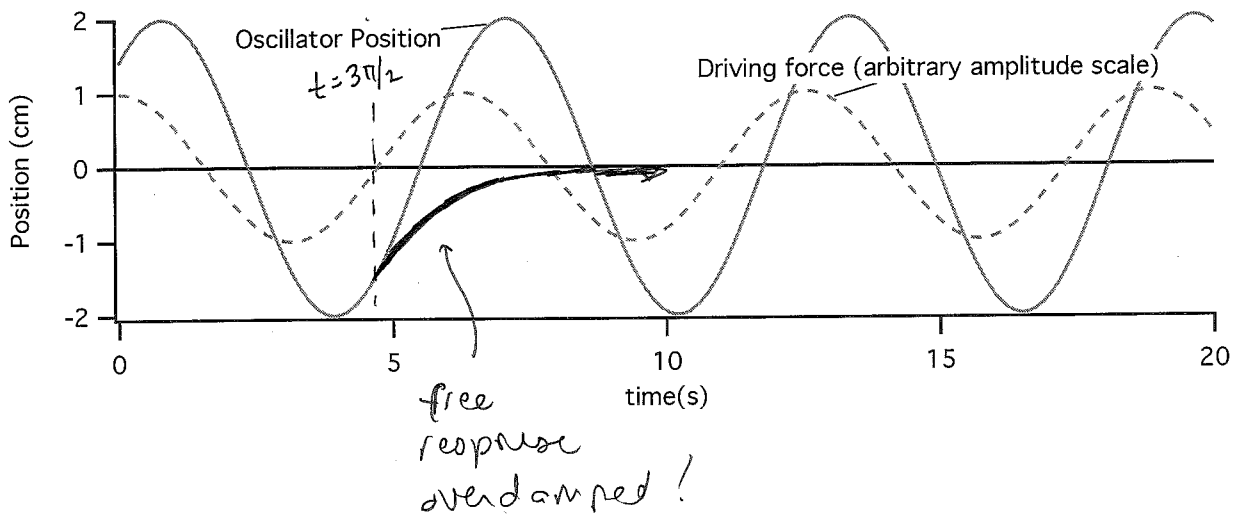
$$= +\frac{2}{\alpha\beta}$$

$$= \frac{2m}{b} \sqrt{v_0}$$

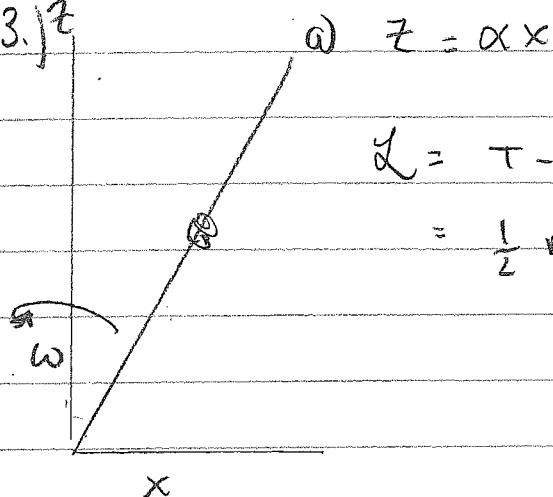
2a) $\delta = \pi/4, < \pi/2. \omega < \omega_0.$

b) $\frac{\pi}{4} = \arctan\left(\frac{2\beta\omega}{\omega_0^2 - \omega^2}\right)$ so $\frac{2\beta\omega}{\omega_0^2 - \omega^2} = 1$

$\frac{2}{\omega_0^2 - 1} = 1 \quad \omega_0^2 = 3 \quad \omega_0 = \sqrt{3}$



3.12



a) $z = \alpha x$

$$\mathcal{L} = T - U$$

$$= \frac{1}{2} m (\dot{x}^2 + \alpha^2 \dot{x}^2) + \frac{1}{2} m x^2 \omega^2 - mg \alpha x$$

b)
$$\frac{\partial \mathcal{L}}{\partial x} = \frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \dot{x}}$$

$$x \omega^2 - g \alpha = \frac{d}{dt} (\dot{x} + \alpha^2 \dot{x}) = (1 + \alpha^2) \ddot{x}$$

c) ΣF_m when $x \omega^2 - g \alpha = 0 \quad x = \frac{g \alpha}{\omega^2}$

If $x \uparrow$, \ddot{x} is +, so unstable.

d) $(1 + \alpha^2) \ddot{x} - \omega^2 x = -g \alpha$

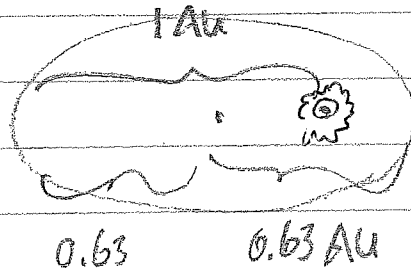
Homogeneous. $x = e^{\pm \beta t} \quad \beta = \frac{\omega}{\sqrt{1 + \alpha^2}}$

Inhom. $x = \frac{+g \alpha}{\omega^2}$

Complete $x = A_1 e^{\beta t} + A_2 e^{-\beta t} + \frac{g \alpha}{\omega^2}$

$$4. a) \left(\frac{T}{T_c} \right)^2 = \left(\frac{1}{2} \right)^2 = \left(\frac{a}{a_c} \right)^3 \quad a = 0.63 a_c$$

Closest approach =
 $2 \times 0.63 - 1 = 0.26 \text{ AU}$



$$b) \text{ Aphelion} = \frac{c_{sc}}{1 - \epsilon}$$

$$\text{ratio} = \frac{1 + \epsilon}{1 - \epsilon} = \frac{1 \text{ AU}}{0.26 \text{ AU}} = 3.85$$

$$\text{Perihelion} = \frac{c_{sc}}{1 + \epsilon}$$

$$1 + \epsilon = 3.85 - 3.85\epsilon$$

$$\epsilon = 2.85 / 4.85 = 0.59$$

$$c) \frac{c}{1} = \frac{c_{sc}}{1 - 0.59} \quad c_{sc} = 0.41 c$$

$$d) \frac{c_{sc}}{c} = \frac{l^2}{l_0^2} \quad l = \sqrt{0.41} l_0 = 0.64 l_0$$

e) Since $\vec{l} = \vec{r} \times \vec{p}$ and r is constant during thrust,
 v must be $= 0.64 v_0$

$$f) \Delta v = 0.36 v_c = 10.7 \text{ km/s}$$

$$v_{ex} = 3 \text{ km/s.}$$

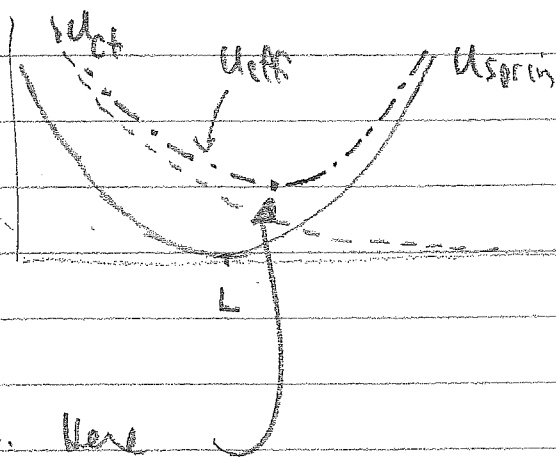
Rocket eqn: $\Delta v = v_{ex} \ln \left(\frac{M_0}{M} \right) \quad \ln \frac{M_0}{M} = \frac{10.7}{3} = 3.528$

$$\frac{M_0}{M} = 35.8 \quad 97\% \text{ fuel.}$$

5. a) $\vec{p} = m\vec{v}$

b) About C.M. $\vec{l} = \vec{r} \times \vec{p}$ ($|\vec{l}| = \frac{L}{2} \cdot mv$)

c) $U = U + U_{eff} = U + \frac{l^2}{2\mu r^2}$ $U = \frac{1}{2}k(r-L)^2$



d.) Yes. Here

Easier to set $F=0 = -k(r-L) + \frac{l^2}{\mu r^3}$

If $r = (L + \delta)$ we find

$$0 = -k\delta + \frac{l^2}{\mu L^3} \left(1 + \frac{\delta}{L}\right)^3 \approx -k\delta + \frac{l^2}{\mu L^3} \left(1 - \frac{3\delta}{L}\right)$$

$$\left(k\mu L^3 + \frac{3l^2}{L}\right)\delta = l^2$$

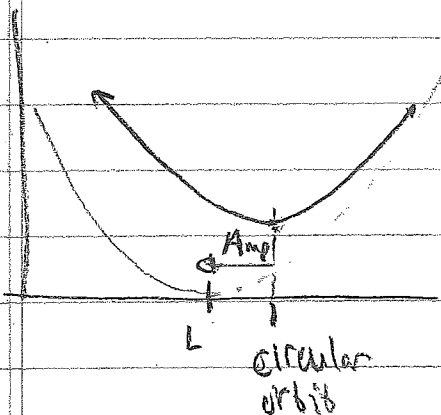
$$\delta = \frac{l^2 L}{k\mu L^4 + 3l^2}$$

Recall $l = \frac{Lmv}{2} = L\mu v$

Can ignore this in small l limit.

$$r_{\text{sup}} = L + \delta$$

5e) The initial kick has $\dot{r}_i = 0$ (it's \perp to r)
 So the amplitude is just the initial amplitude



$$A = \delta.$$

f) Both U_{spring} and U_{rot} are concave up. Their sum will thus be ever steeper upward, and $\omega \uparrow$. $\omega > \sqrt{\frac{k}{m}}$

g) $k_{eff} = -\frac{d^2F}{d\delta^2}$ (this is also the 2nd derivative of U_{eff})

$$= -\left(-k - \frac{3l^2}{\mu L^4}\right) = k + \frac{3l^2}{\mu L^4} \quad \omega = \sqrt{\frac{k_{eff}}{\mu}}$$

h) $\mu = \frac{m_1 m_2}{m_1 + m_2} = \frac{M}{2} = 1 \text{ kg}$ $v = 1 \text{ m/s}$ $l = L \mu v = 1 \cdot 1 \cdot 1 = 1 \text{ kg} \cdot \frac{\text{m}^2}{\text{s}}$

So $k_{eff} = 13 + \frac{3 \cdot 1}{1 \cdot 1} = 16$ $\omega = \sqrt{\frac{16}{1}} = 4 \text{ s}^{-1}$

the whole system rotates with angular frequency $\frac{\omega}{L} = 1 \text{ s}^{-1}$

Orbit:

